

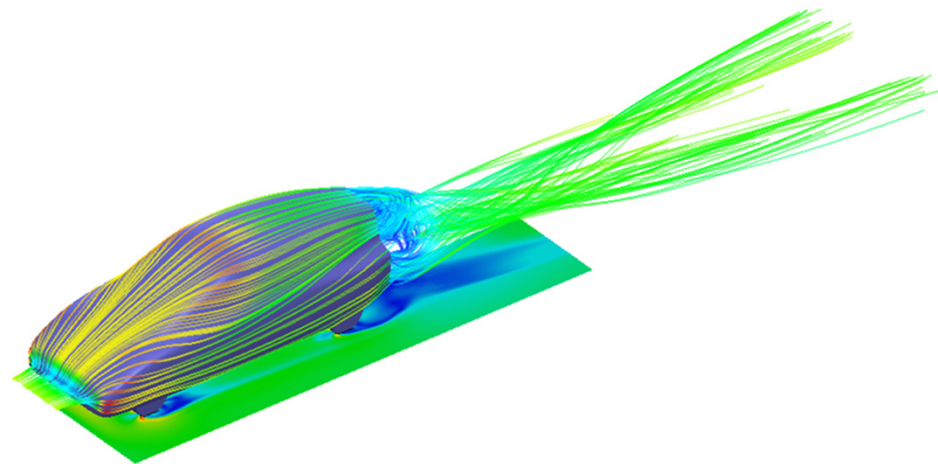
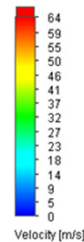
# Comparison of the ASMO Car Model with Experimental Data and Simulations

Mike Gruetzmacher, Mentor Graphics



# Content

- Introduction
- Physical Engineering Models
- The ASMO example
- Conclusion

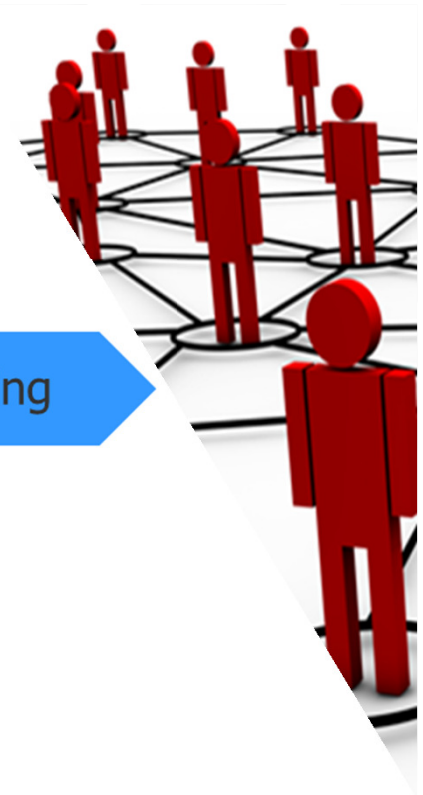
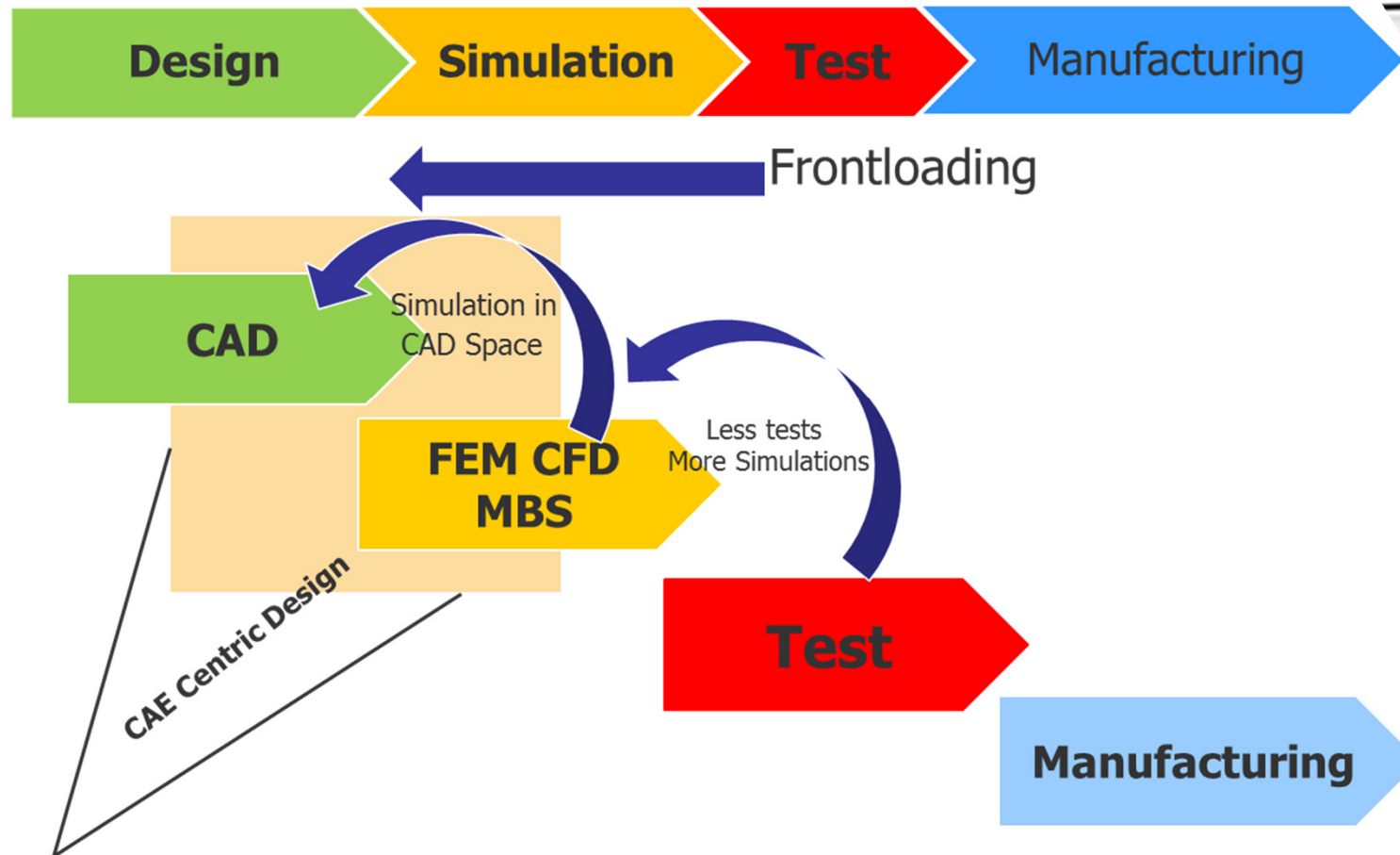


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# Introduction



# Introduction

- Early development phase
- Aerodynamics car analysis
  - Wind tunnel experiments
  - CFD analysis
- Increasing flexibility, cost reduction, design modifications
- Impact of components (e.g. radiator grill, rear wing, mirrors)



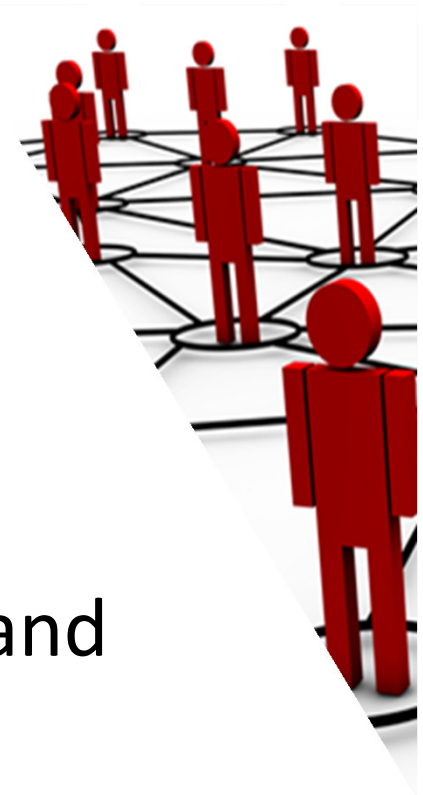
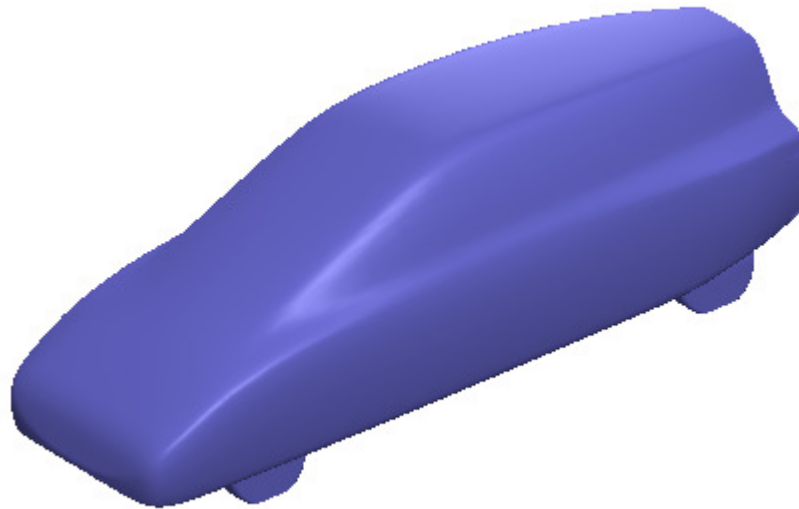
# Introduction

- Real car models: very complex geometry
- Simplified reference bodies for
  - Basic investigations
  - Code validations



# Introduction

- ASMO example: „Aerodynamisches Studienmodell“
- Wind tunnel tests by Daimler Benz and Volvo



# Content

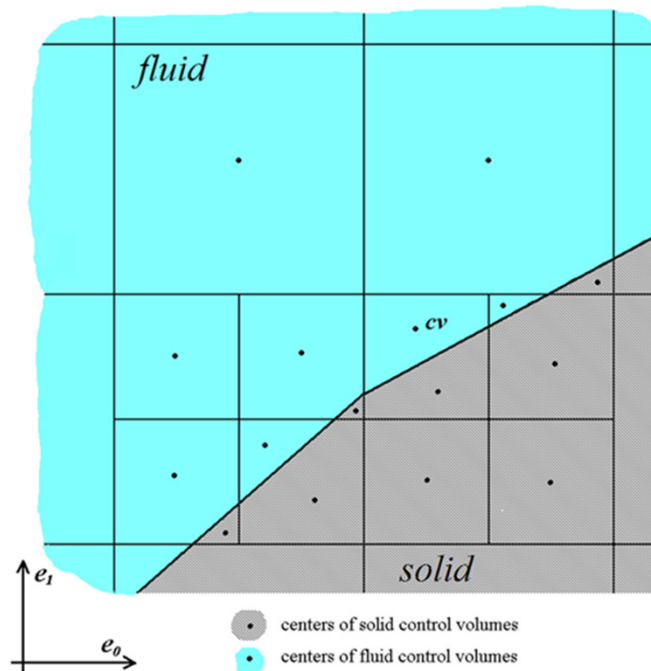
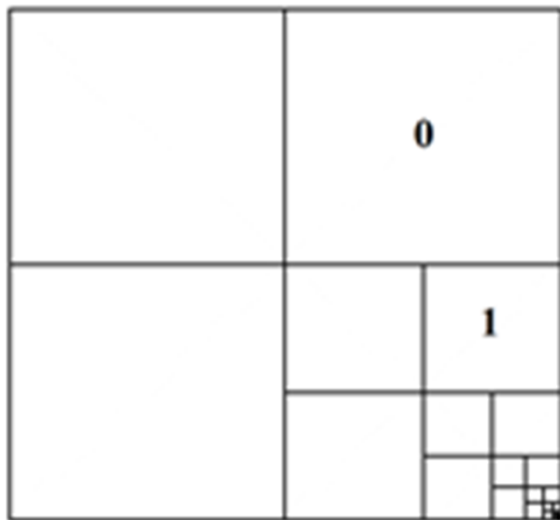
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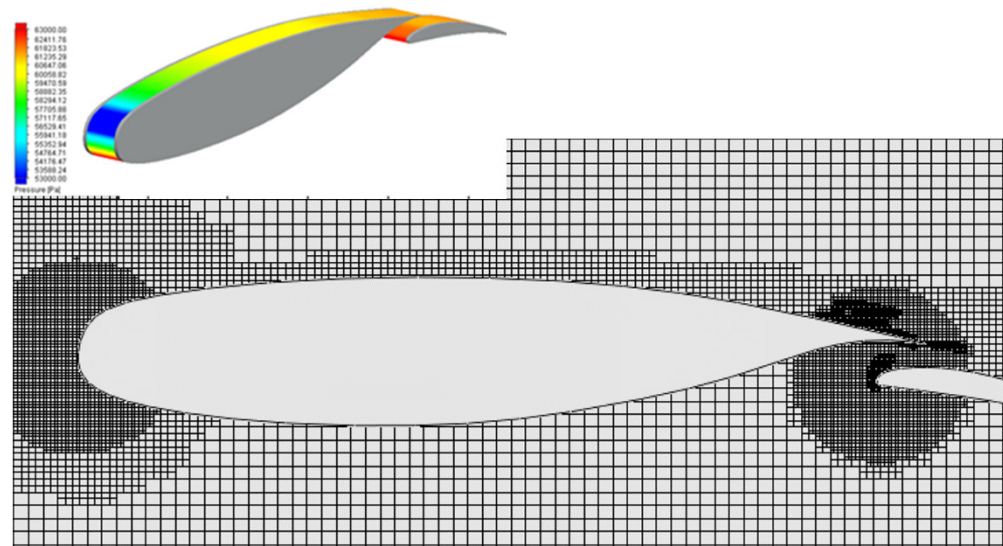
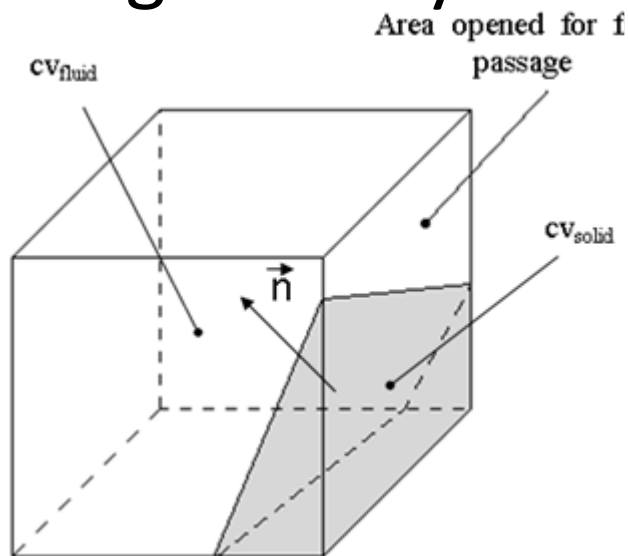
# Meshing

- Mesh: Rectangular Grid
- Refinements by cutting parallelepipeds



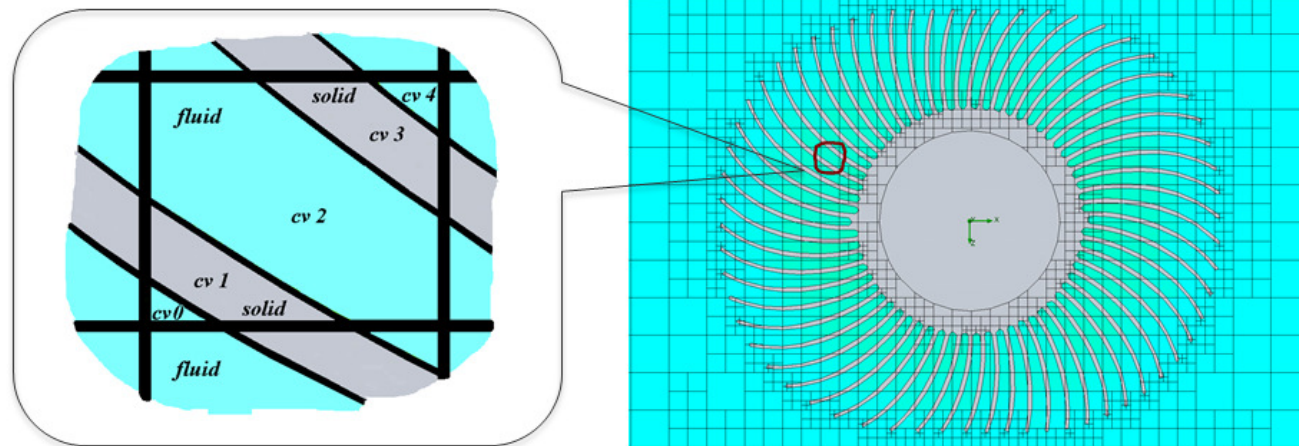
# Meshing

- Model body splits the cell into two or more sub-control volumes (polyhedrons) to represent the geometry



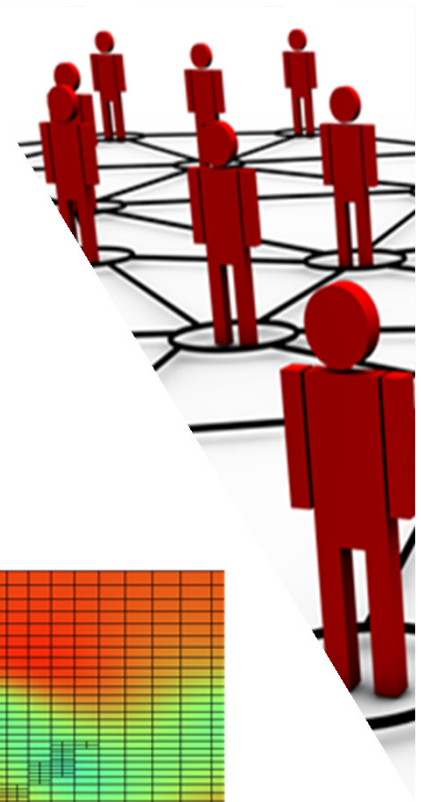
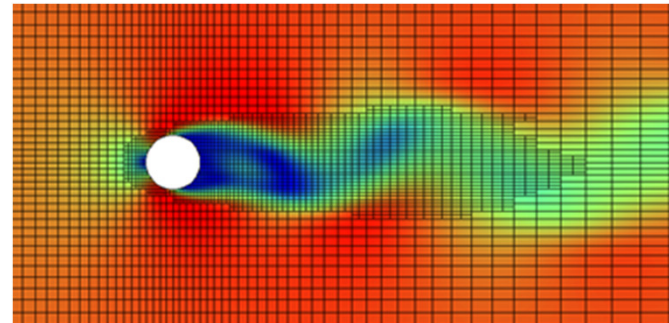
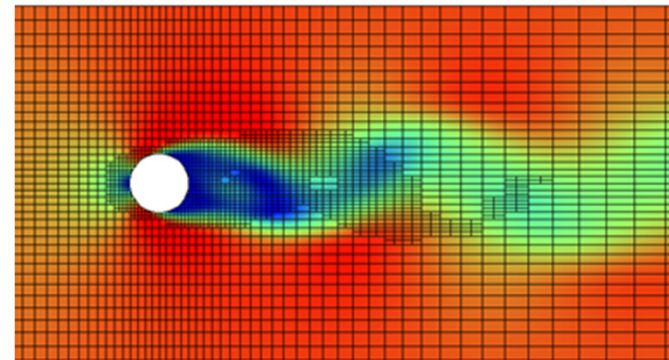
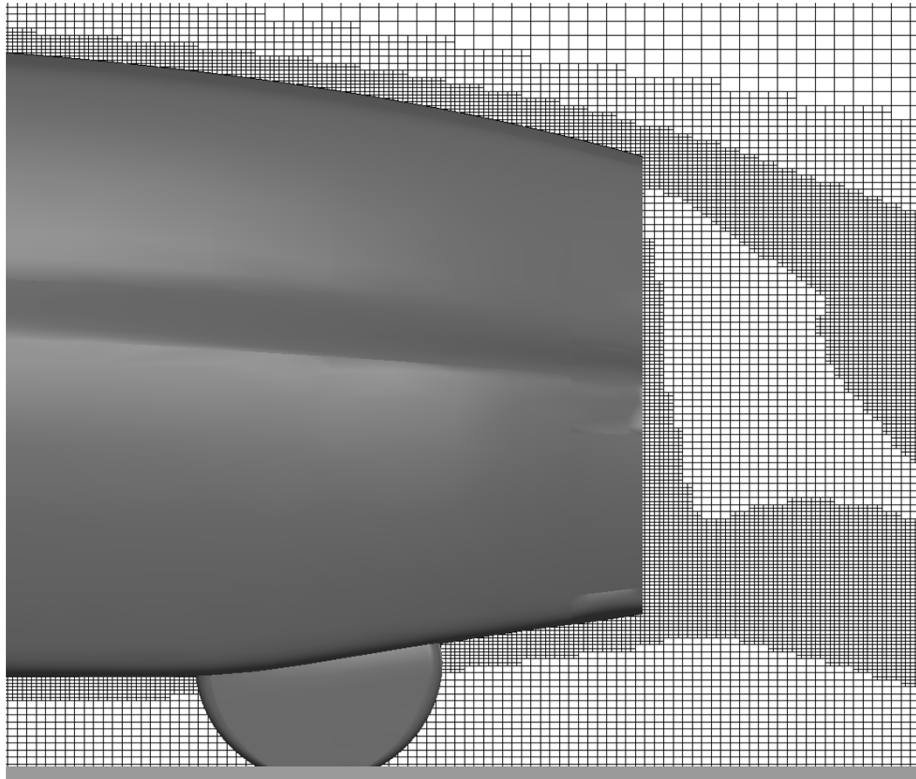
# Meshing

- Several sub-control volumes for very small features, f.i. thin layers



# Meshing

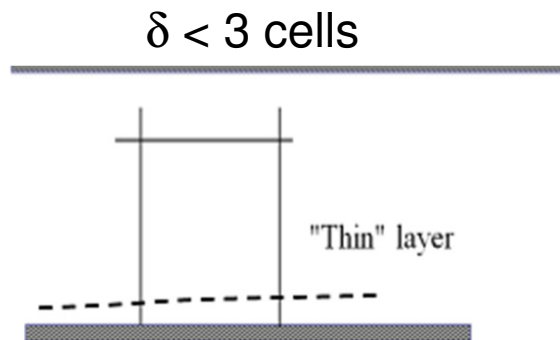
- Solution adaptive refinement for regions with high flow gradients





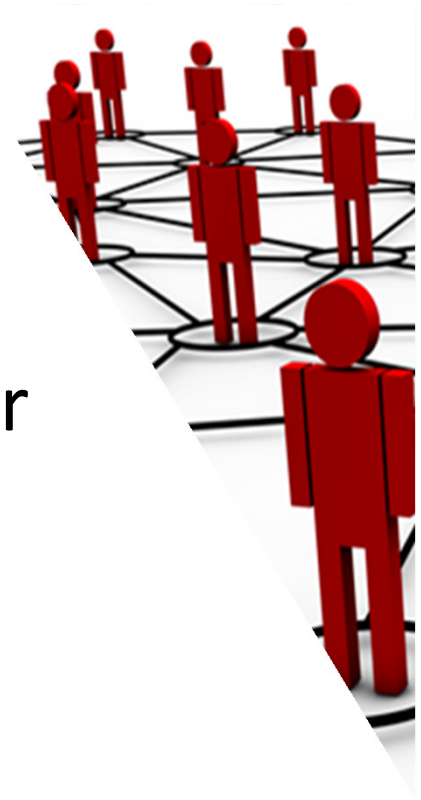
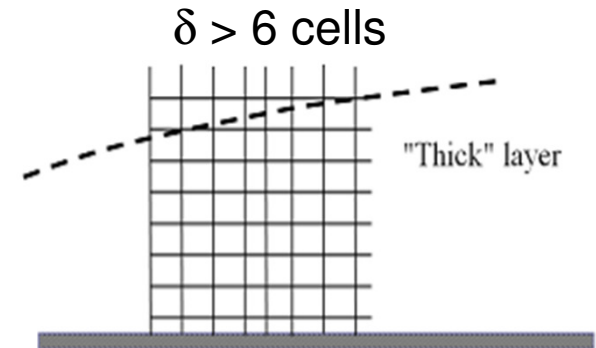
# Boundary Layer

- FloEFD separates the boundary layer treatment into:
  - „Thin“ Boundary Layer
  - „Thick“ Boundary Layer



$3 < \delta < 6$  cells

Interpolated  
between both  
Types automatically



# Physical Engineering Models

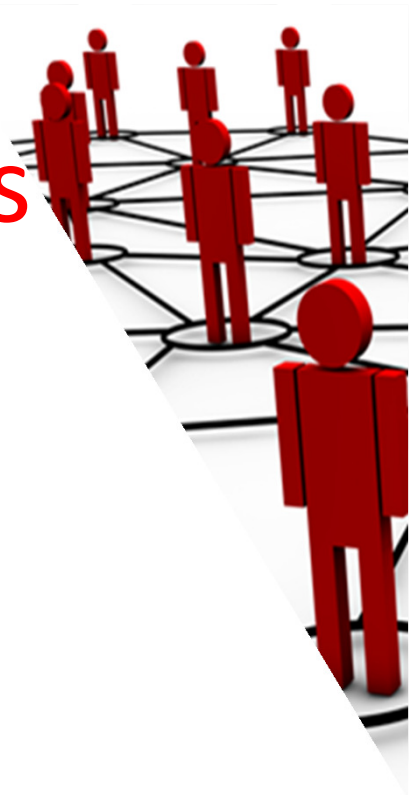
- One modified k- $\varepsilon$  turbulence model

$$\frac{\partial \rho k}{\partial t} + \frac{\partial}{\partial x_i} (\rho u_i k) = \frac{\partial}{\partial x_i} \left( \left( \mu + \frac{\mu_t}{\sigma_k} \right) \frac{\partial k}{\partial x_i} \right) + S_k$$

$$\frac{\partial \rho \varepsilon}{\partial t} + \frac{\partial}{\partial x_i} (\rho u_i \varepsilon) = \frac{\partial}{\partial x_i} \left( \left( \mu + \frac{\mu_t}{\sigma_\varepsilon} \right) \frac{\partial \varepsilon}{\partial x_i} \right) + S_\varepsilon$$

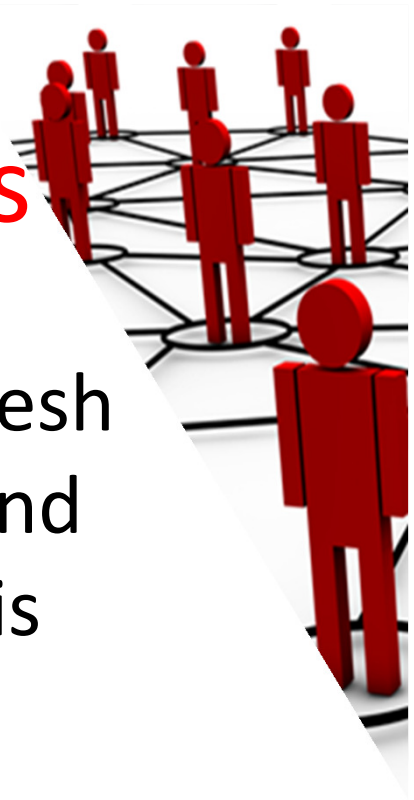
$$S_k = \tau_{ij}^R \frac{\partial u_i}{\partial x_j} - \rho \varepsilon + \mu_t P_B$$

$$S_\varepsilon = C_{\varepsilon 1} \frac{\varepsilon}{k} \left( f_1 \tau_{ij}^R \frac{\partial u_i}{\partial x_j} + \mu_t C_B P_B \right) - C_{\varepsilon 2} f_2 \frac{\rho \varepsilon^2}{k}$$



# Physical Engineering Models

- Algebraic solver is adapted to the mesh and the boundary layer treatment and includes the turbulence model in this approach
- high convergence stability and accuracy



# Physical Engineering Models

- Advanced Meshing Technology in FloEFD™

<http://go.mentor.com/2gogl>

- Enhanced Turbulence Modelling in FloEFD™

<http://go.mentor.com/2glzd>





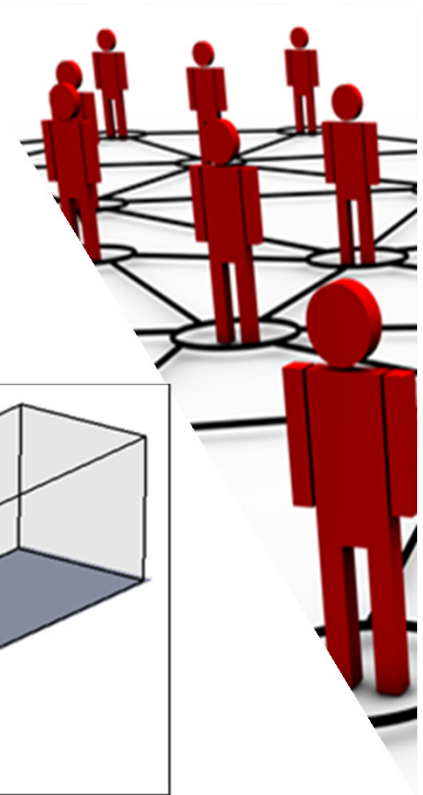
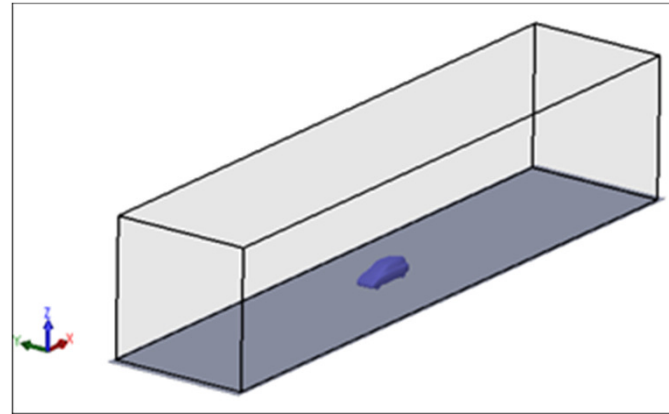
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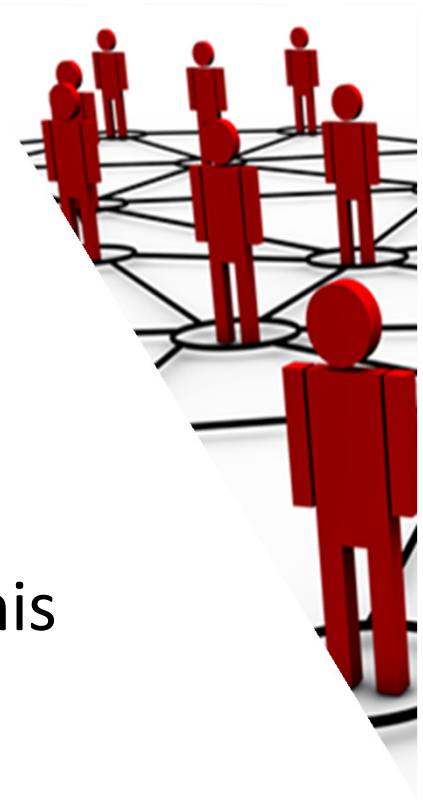
# The ASMO example

- Boundary Conditions
  - Velocity 50 m/s
  - Medium Air
  - Temperature 20 °C
  - Pressure 101,325 Pa
  - Reynolds  $Re=2,700,000$  (vehicle length as reference)
- No movement applied to the floor surface (as in experiment)



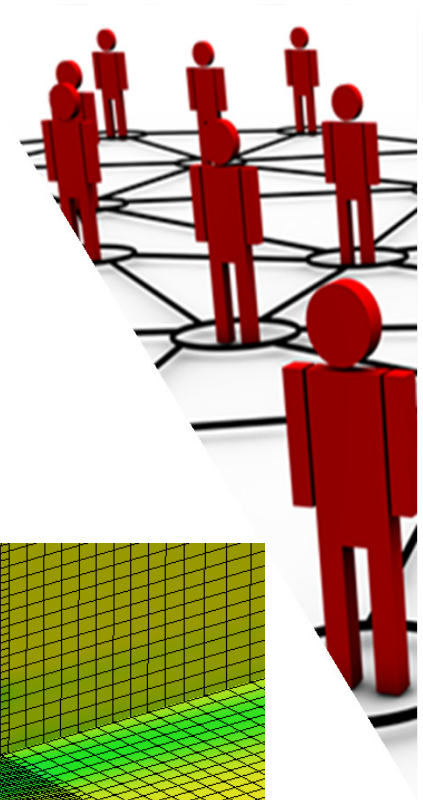
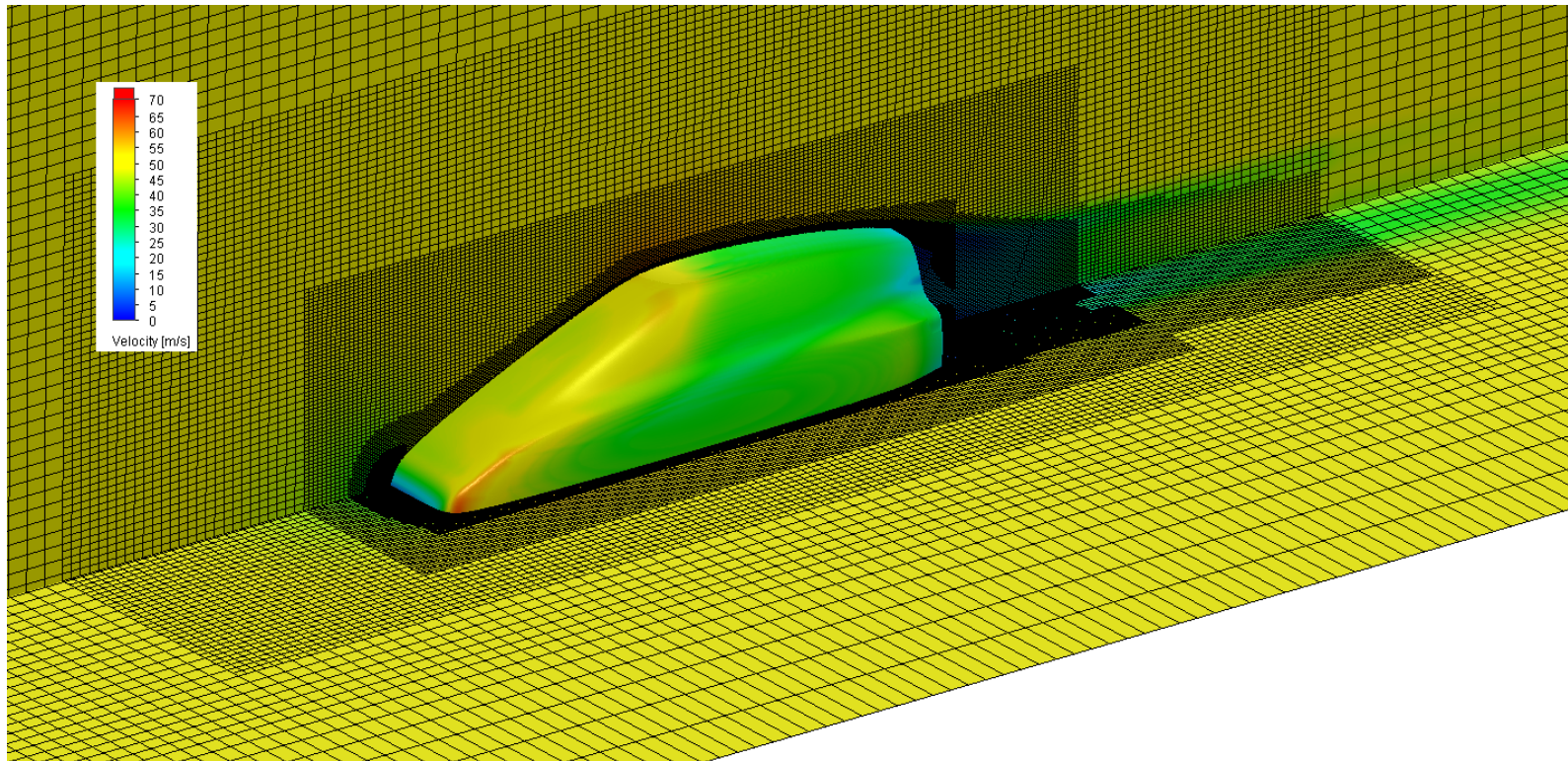
# The ASMO example

- Mesh:
  - Solution adaptive refinement used (two refinements are performed for this calculation)
- Number of cells:
  - Initial: 3.6 M
  - After the 2<sup>nd</sup> refinement: 13.5 M



# The ASMO example

- Calculated Mesh:



# The ASMO example

- Time Overhead Across the Simulation Workflow

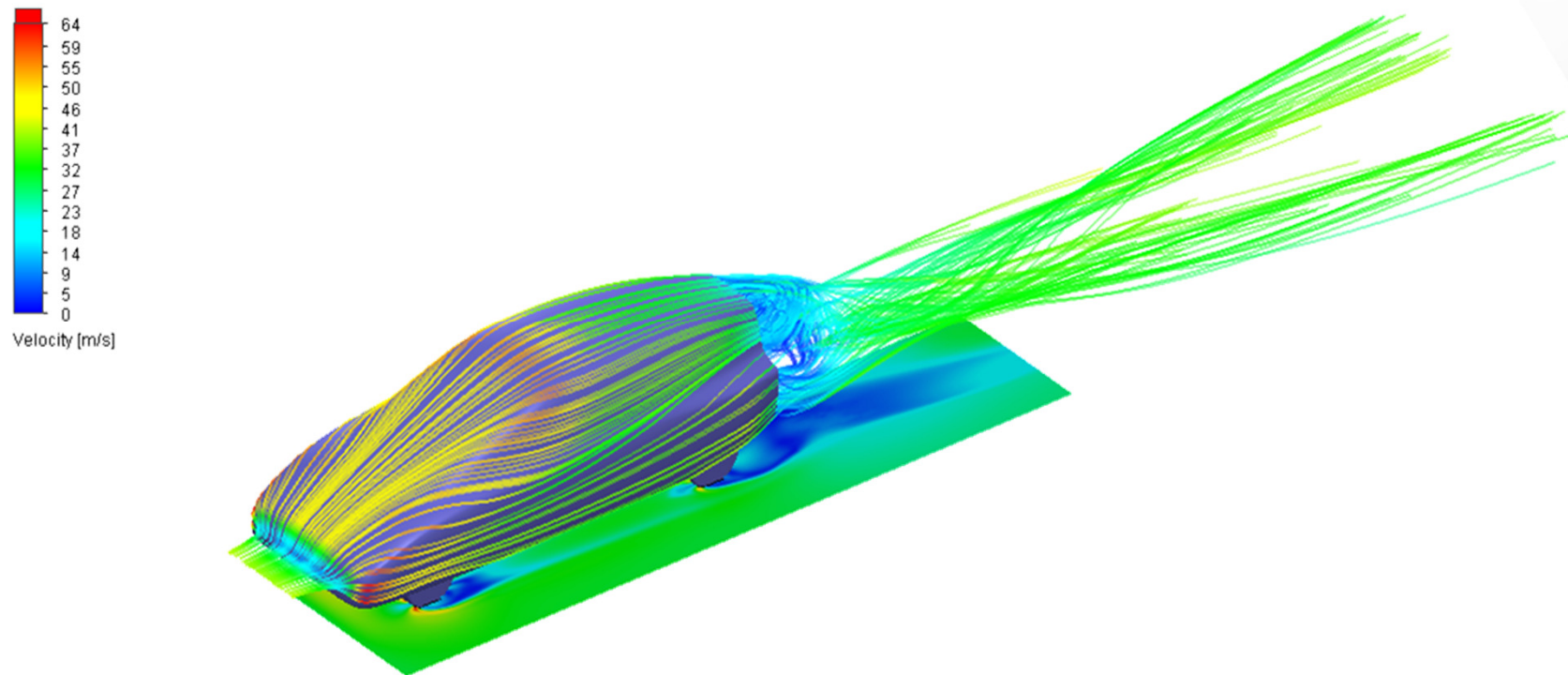
Time Overhead Across the Simulation Workflow	in h
Setup Calculation project definition, pre-processing	1
Solving time (including meshing and refinements)	18*
• Initial 1st automatic meshing time	0,03 (1,5 min)
Results processing	4
<b>Total</b>	<b>23</b>
User time, approximately	5

\* Intel(R) Xeon(R) CPU E5-2630 v3 @ 2.40GHz (2 CPU)



# The ASMO example

## Result discussion





# The ASMO example

- Drag Coefficient

$$C_d = \frac{2F_d}{\rho u^2 A}$$

$F_d$  = drag force in flow direction

$\rho$  = fluid density

$u$  = fluid velocity

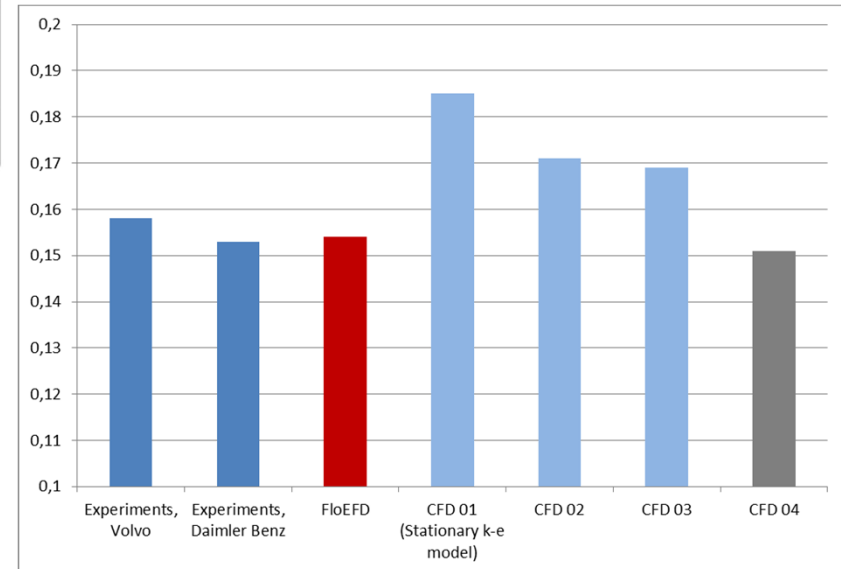
$A$  = projected surface



# The ASMO example

- Results: Drag Coefficient

Drag coefficient	
Experiments, Volvo	0,158
Experiments, Daimler Benz	0,153
FloEFD	0,154
CFD 01 (Stationary k-e model)	0,185
CFD 02	0,171
CFD 03	0,169
CFD 04	0,151





# The ASMO example

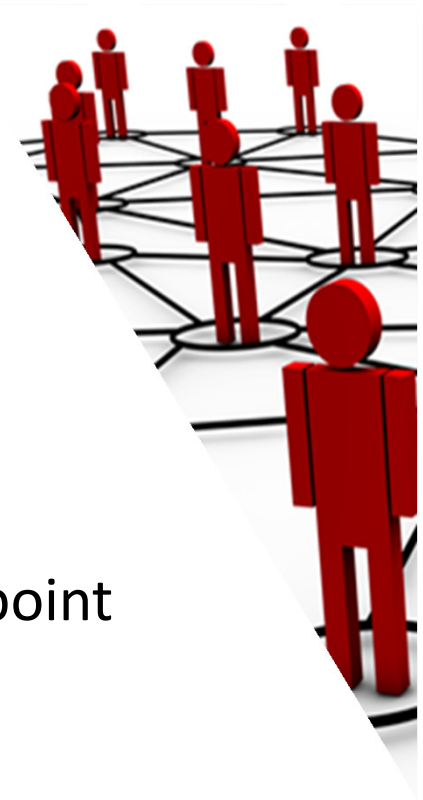
- Pressure Coefficient  $C_P = \frac{p - p_\infty}{\frac{1}{2} \rho_\infty v_\infty^2}$

$p$  = pressure at the corresponding point

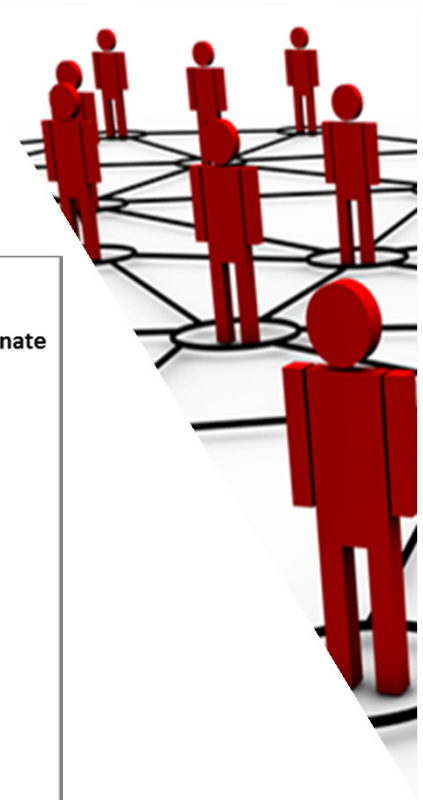
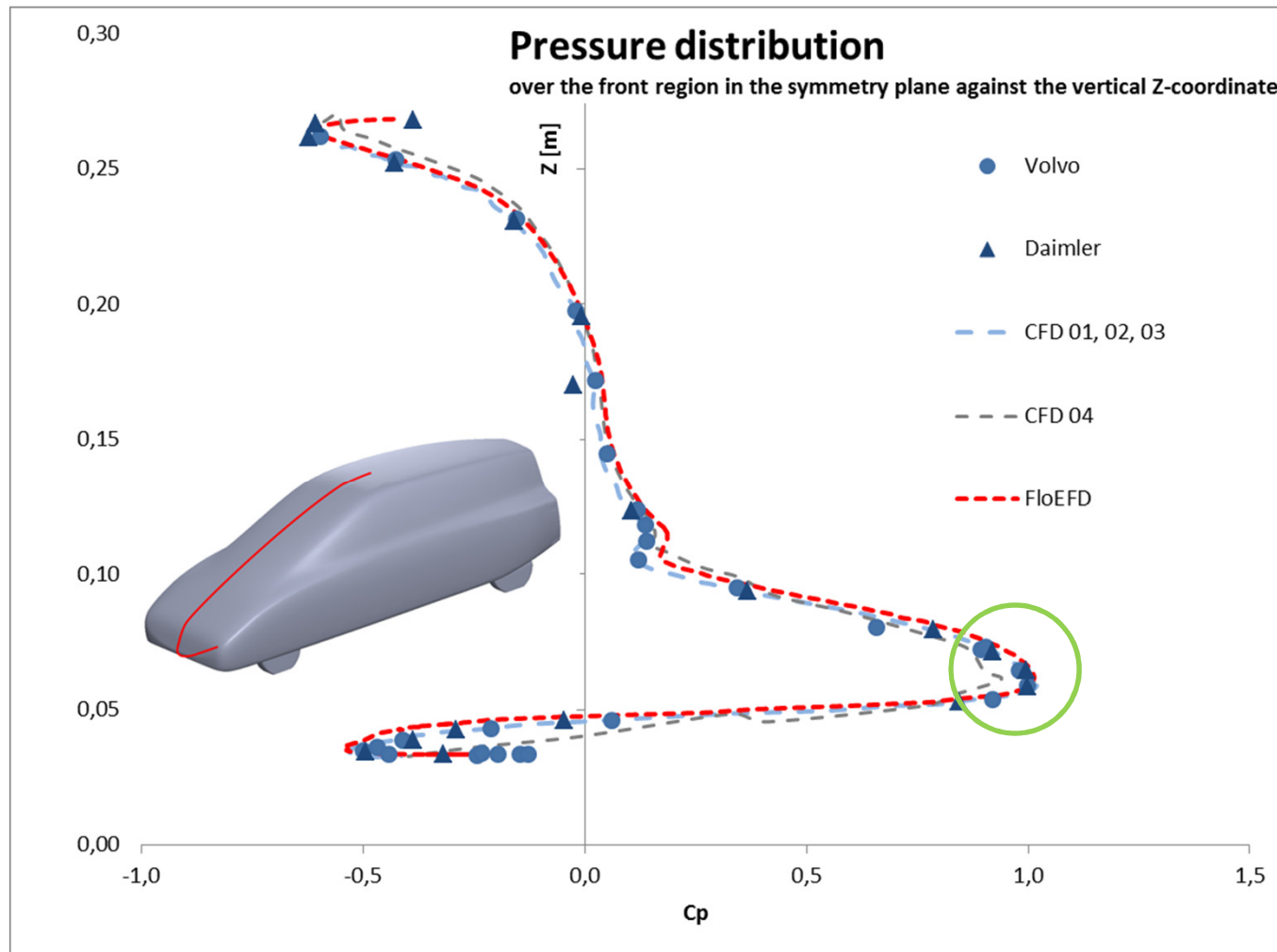
$p_\infty$  = pressure in the freestream

$\rho_\infty$  = fluid density in the freestream

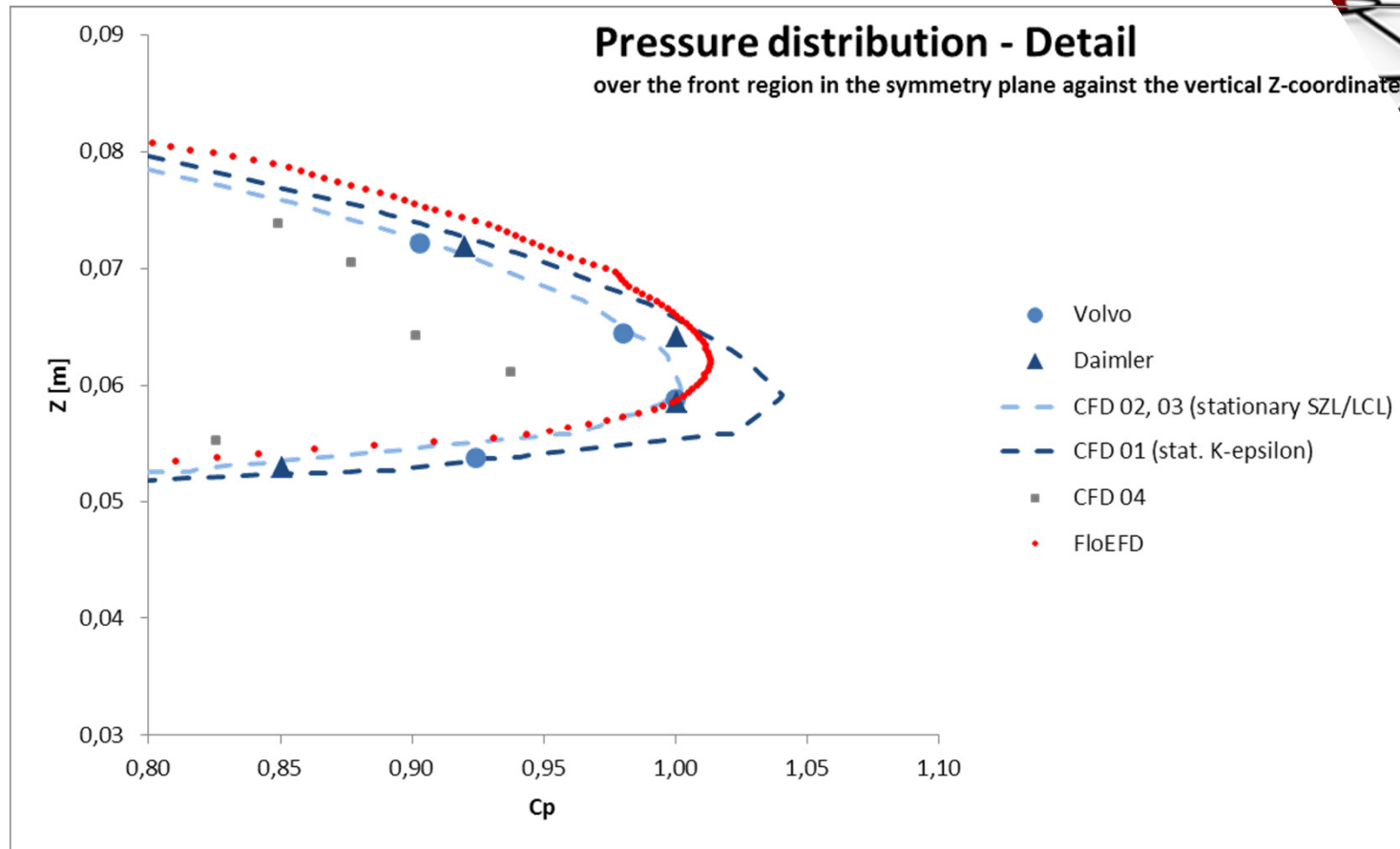
$v_\infty$  = freestream velocity



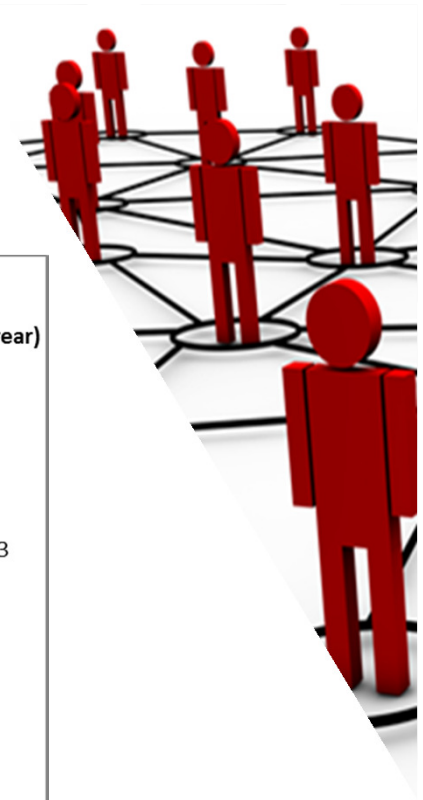
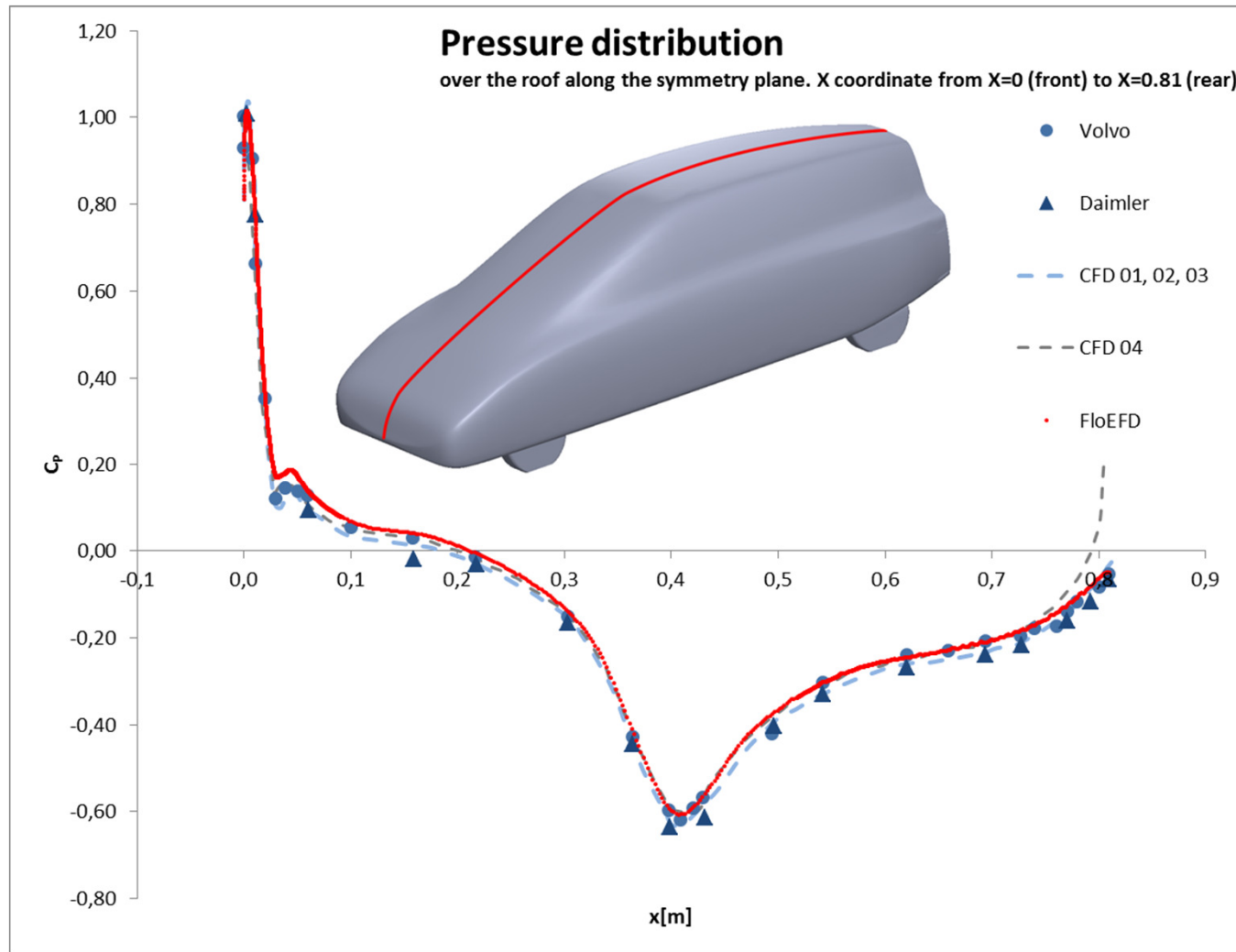
# The ASMO example



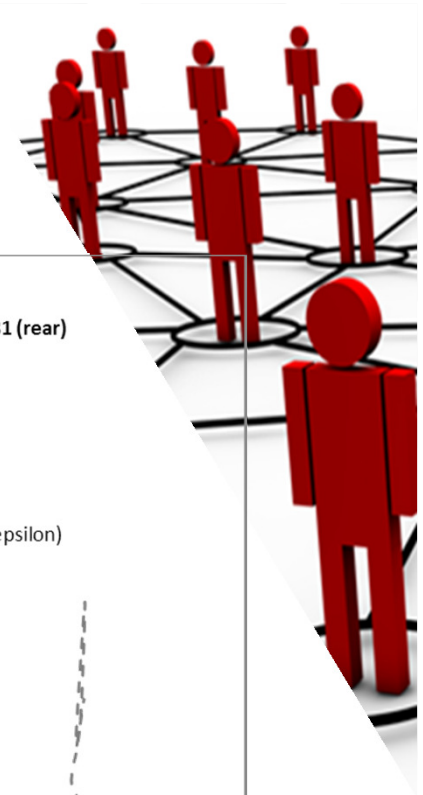
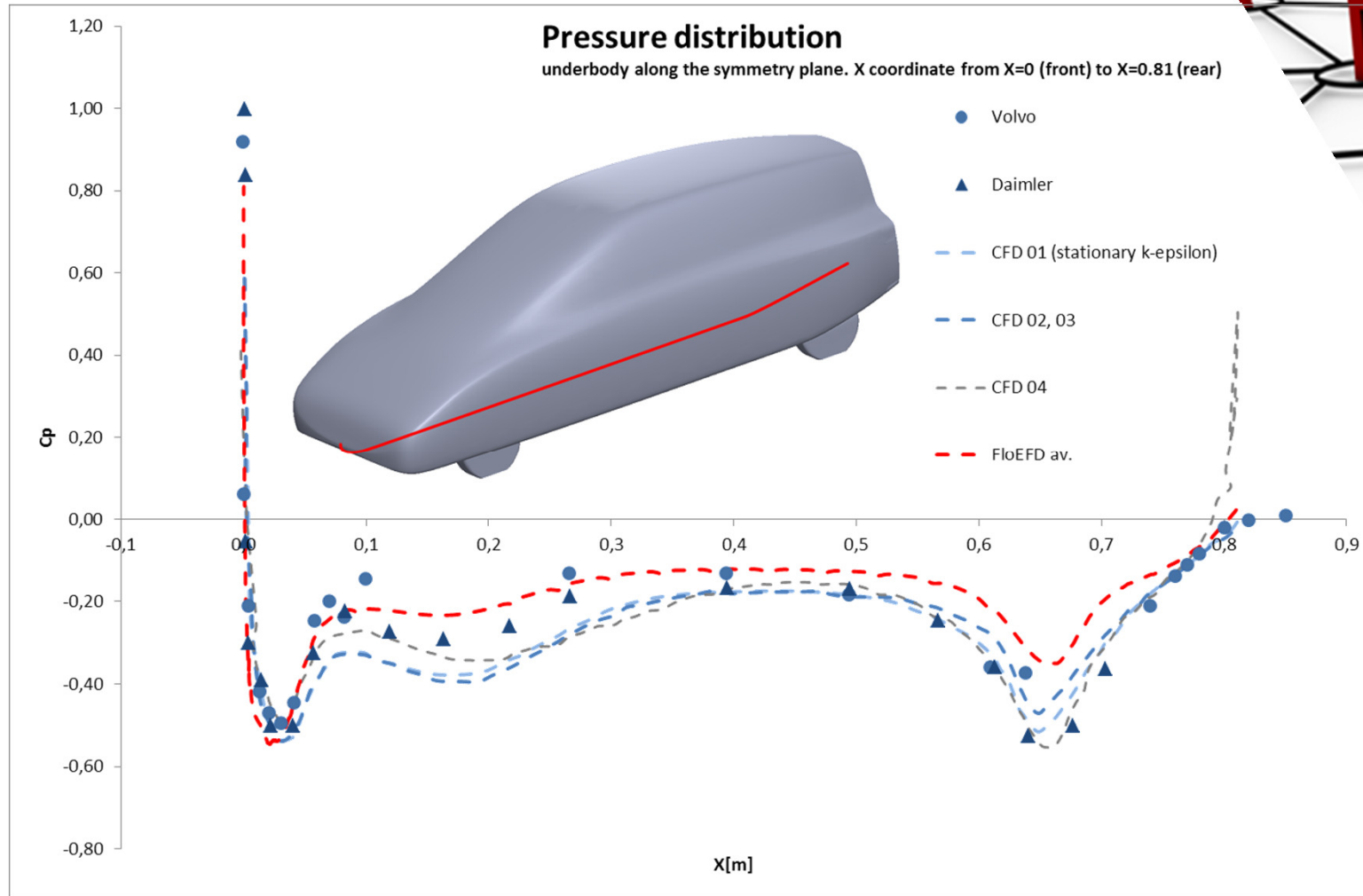
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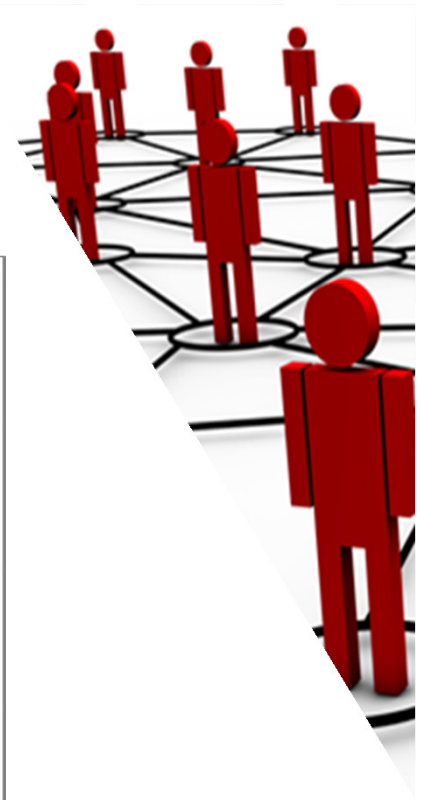
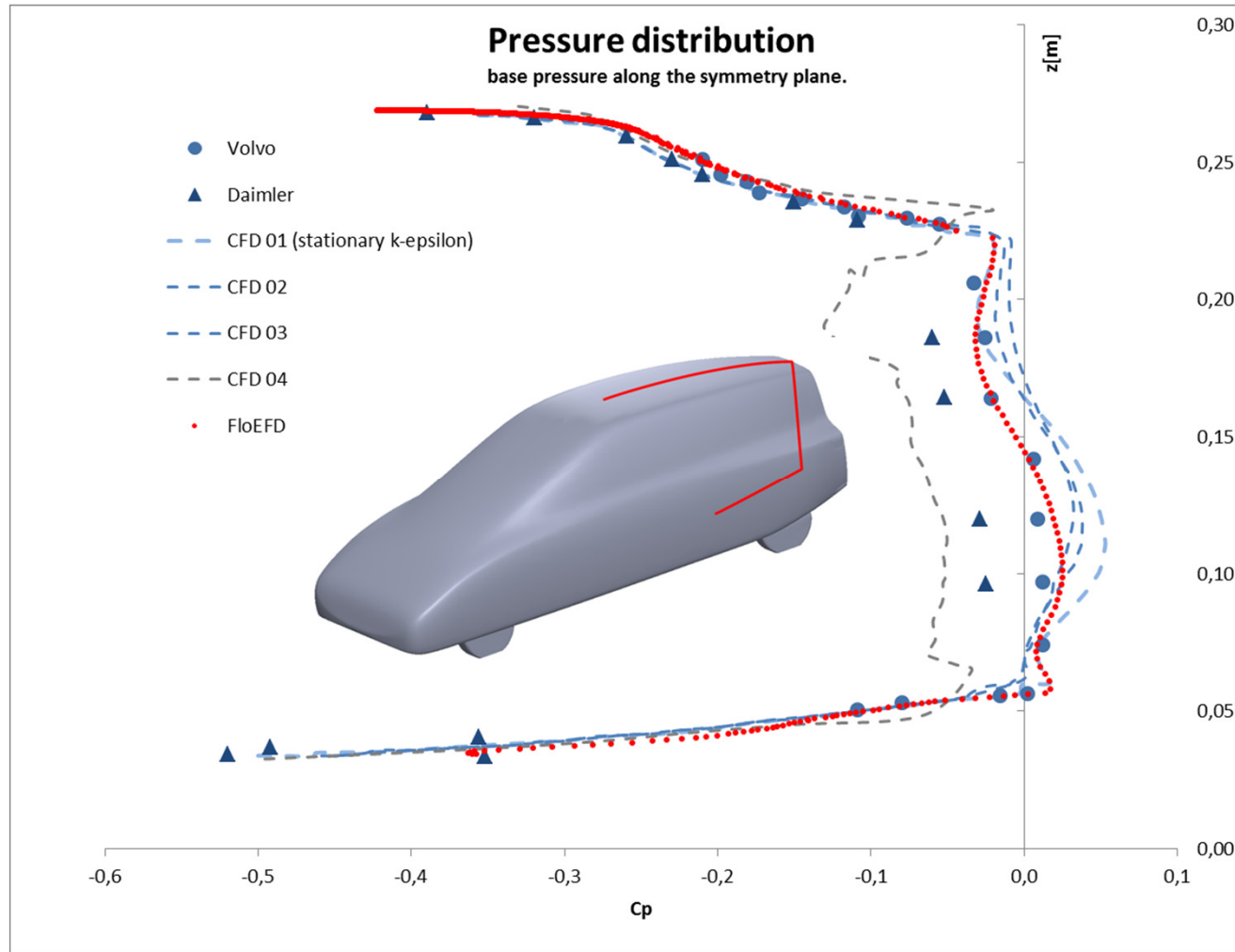
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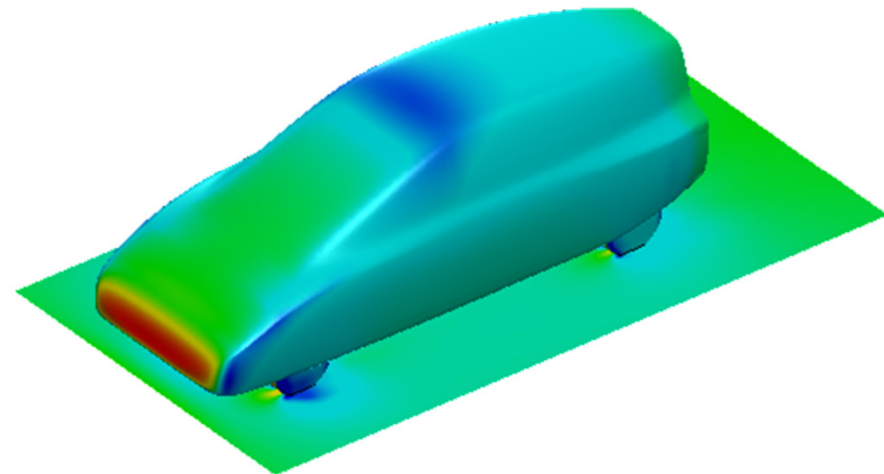
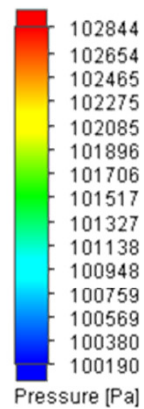
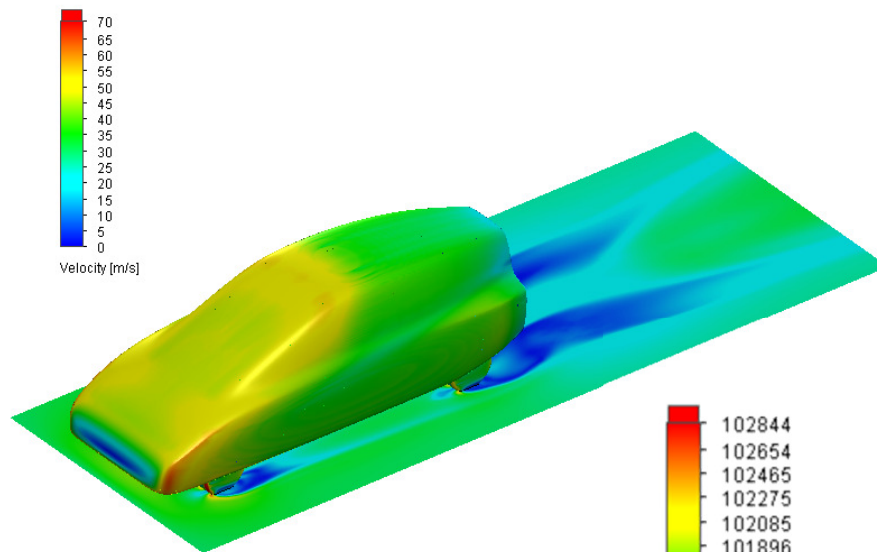
# The ASMO example



# The ASMO example



# The ASMO example





# Content

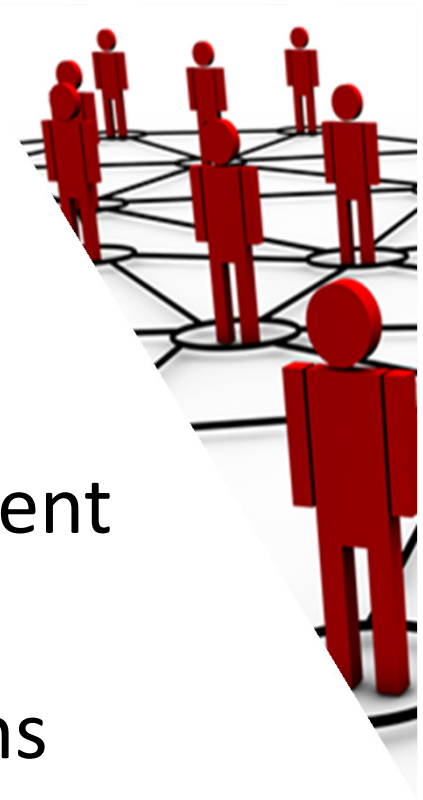
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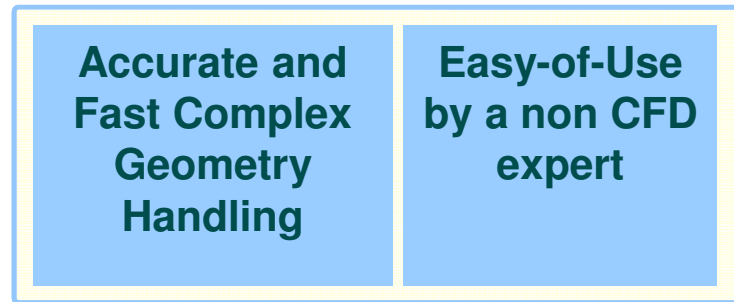


# Conclusion

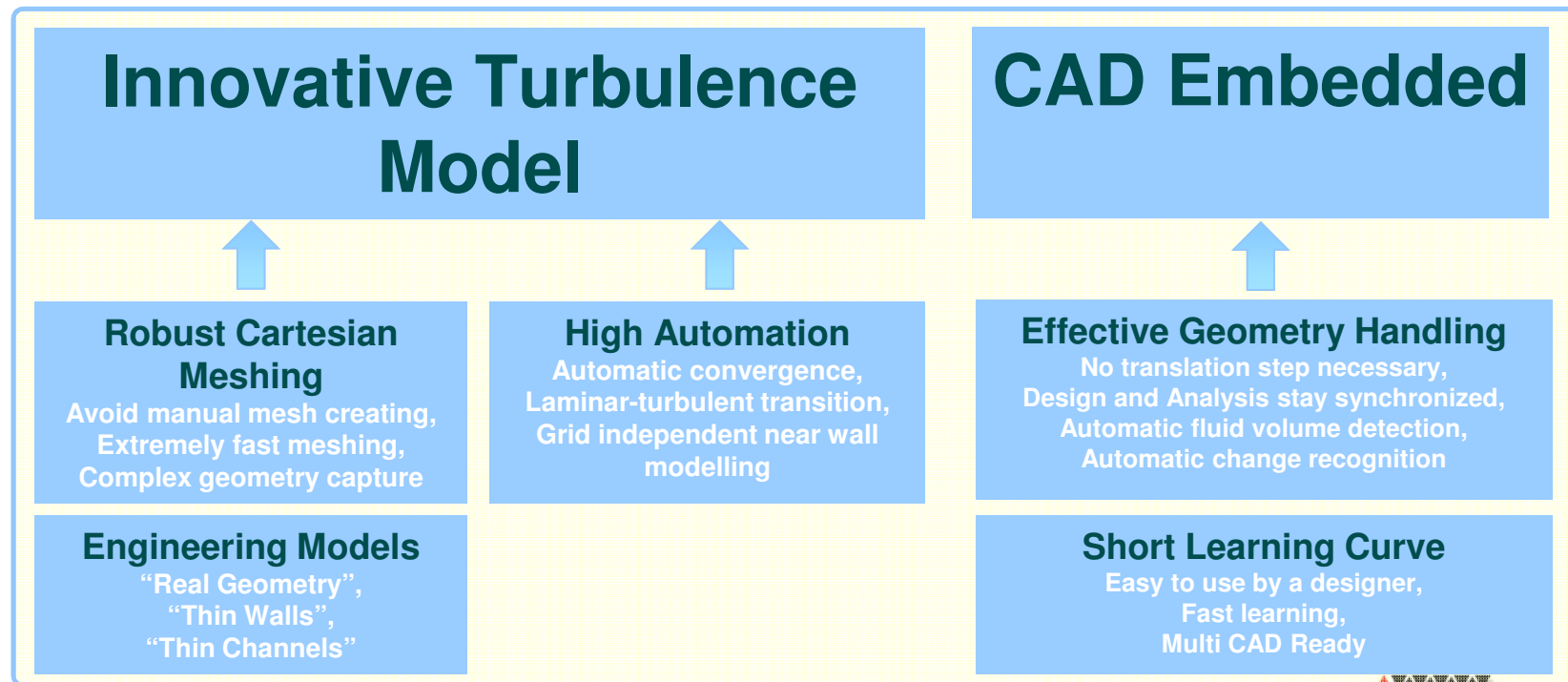
- Good agreement for the drag coefficient and the pressure coefficient curves
- Strong pressure coefficient variations due to turbulent wake influencing the diffuser area
- The CAD embedded CFD tool FloEFD shows reliable results



# Conclusion



**Frontloading**



# Conclusion

NAFEMS European Conference:

## Computational Fluid Dynamics (CFD) – Beyond the Solve

The complete CFD activity chain across all industries from geometry to final conclusions – best practice methods and tools, automation, optimisation, verification and validation.

2 - 3 December, Munich, Germany

- CAD embedded simulation enables “Frontloading”
- Same level of accuracy compared to traditional, well established CFD tools and experimental results



# Thanks for your attention !



## References

- [1] Whitepaper “Enhanced Turbulence Modeling In FloEFD”, Mentor Graphics Corporation, 2011, <http://go.mentor.com/2glzd>
- [2] Perzon S., Davidson L.: "ON TRANSIENT MODELING OF THE FLOW AROUND VEHICLES USING THE REYNOLDS EQUATION", ACFD 2000 Beijing, 2000
- [3] <http://www.xflowcfcd.com/pdf/Project01-ASMO.pdf>, "Aerodynamic analysis involving moving parts with XFlow", 2010
- [4] Marovic B., “CAD Embedded CFD vs. Traditional CFD Codes in a Blind JSAE Benchmark”, NAFEMS World Congress 2015